

# NASA's Habitation Development Status: Current Concepts and ISRU Opportunities

National Aeronautics and  
Space Administration



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NASA's Habitation Systems Development Office

# NASA's Habitation Systems Development Team



NASA Marshall Space Flight Center – Habitation Systems Development Office



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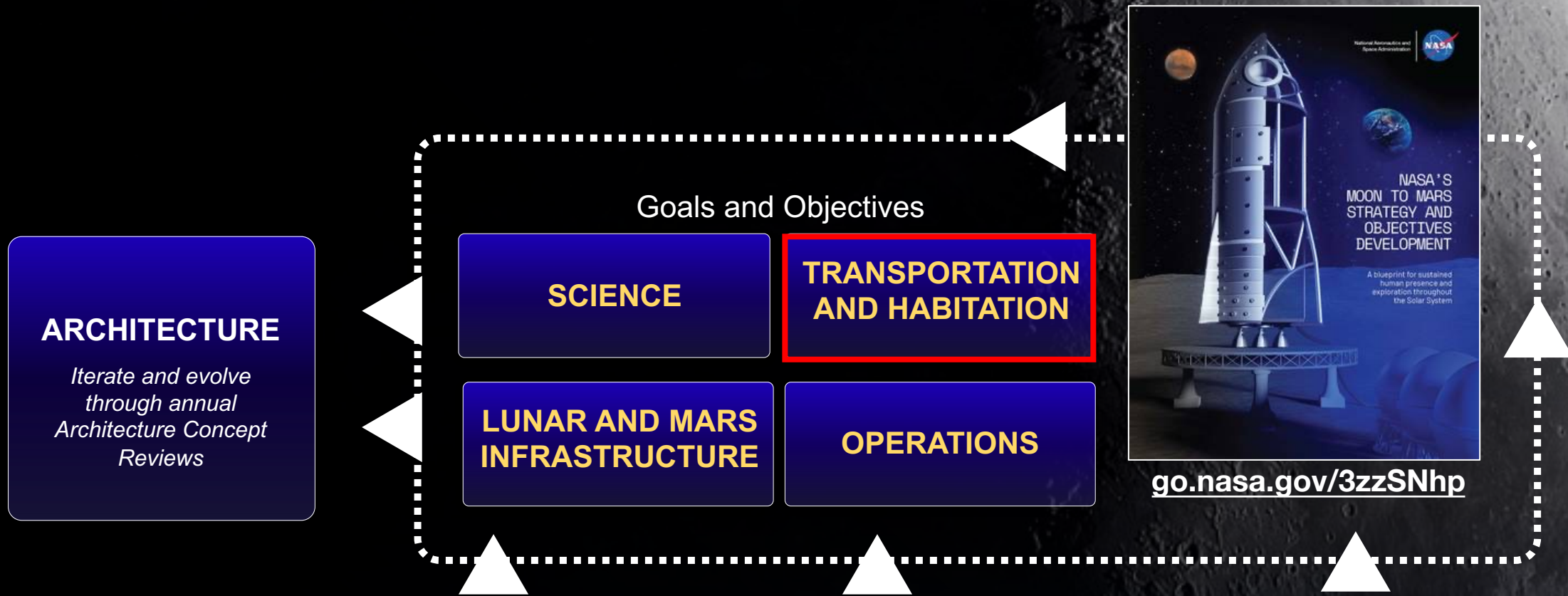
**James E. Johnson\***  
Habitation Systems Engineer

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# NASA's Moon to Mars Strategy and Objectives

A blueprint for future human exploration (Architecting from the Right)



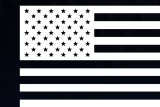
Requested feedback on these objectives in summer 2022 from the following key stakeholders:



NASA workforce:  
our greatest asset



International partners: our key  
current and future, anticipated  
collaborators



U.S. industry, academia, DOE, NIH,  
NSF, etc.: our national leaders in  
space research and capabilities

# Executing from the Left: Segments and Sub-architectures



**Segment:** A portion of the architecture, identified by one or more notional missions or integrated use cases, illustrating the interaction, relationships, and connections of the sub-architectures through progressively increasing operational complexity and objective satisfaction.



## Human Lunar Return

Initial capabilities, systems, and operations necessary to re-establish human presence and initial utilization (science, etc.) on and around the Moon.

**Focus for ACR 22**



## Foundational Exploration

Expansion of lunar capabilities, systems, and operations supporting complex orbital and surface missions to conduct utilization (science, etc.) and Mars forward precursor missions.



## Sustained Lunar Evolution

Enabling capabilities, systems, and operations to support regional and global utilization (science, etc.), economic opportunity, and a steady cadence of human presence on and around the Moon.



## Humans to Mars

Initial capabilities, systems, and operations necessary to establish human presence and initial utilization (science, etc.) on Mars and continued exploration.

**Focus for ACR 23**

**Sub-architecture:** A group of tightly-coupled systems, functions, and capabilities that perform together to accomplish architecture objectives.

Communication, Positioning, Navigation, and Timing •  
Habitation • Human Systems • Logistics • Mobility Systems  
• Power • Transportation • Utilization Systems

**ACR = Architecture Concept Review**



# Surface Habitat Reference Concept

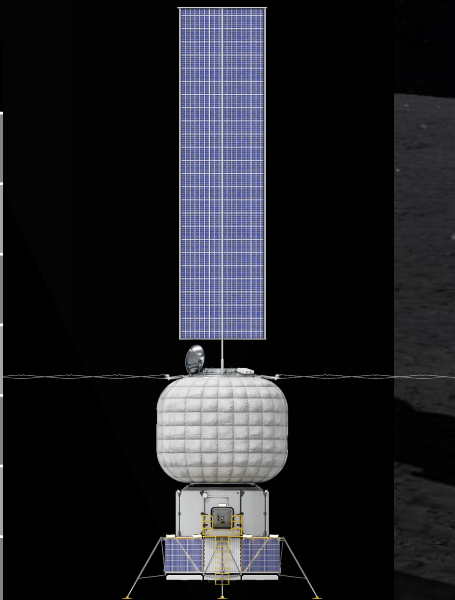
NASA's surface habitat concept serves as a central hub for building & initial sustainment of Artemis' **Foundational Exploration** activities

Key Surface Habitat functions:

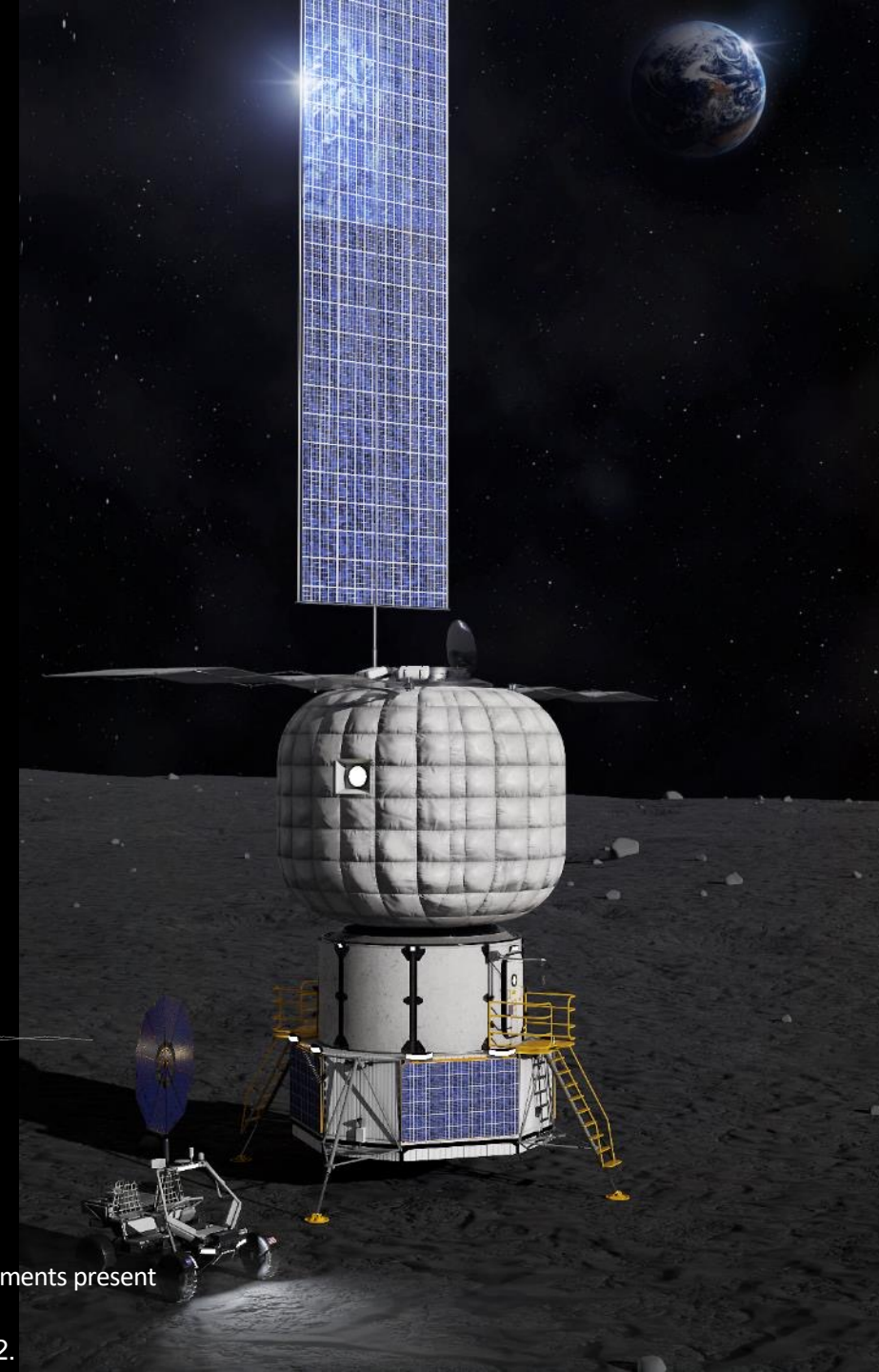
- Power generation & energy storage (100 hrs)
- Open/Regenerative Environmental Control & Life Support (ECLS)
- Extravehicular Activity (EVA) support & suit maintenance
- Communications & data relay
- Medical support

Surface Habitat Concept Design Characteristics<sup>1</sup>

Crew Size	2 crew (nominal) – 4 crew (surge*)
Mission Duration	~30 days (initial) – 60 days (extended)
Lifetime	15 years
Mass Target	12 mt (goal)
Max Power	15 KWe
Pressurized Volume	~190 m <sup>3</sup>
Operating Pressure	8.2 – 10.2 psia (70.3 – 56.5 kPa)
Oxygen Composition	34-26.5% (limits for 8.2-10.2 psia)



\*Surge capacity when other elements present



<sup>1</sup> Harris, Danny W., et al. *Moon to Mars (M2M) Habitation Considerations: A Snap Shot As of January 2022*. No. M-1538. 2022.

# Transit Habitat Reference Concept

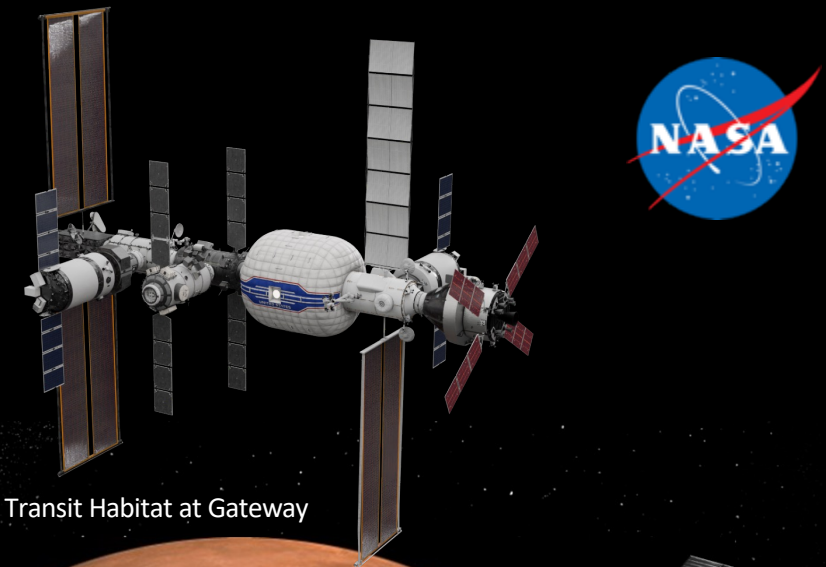
NASA's transit habitat serves as crew living quarters during Mars analog and exploration missions during Artemis' **Foundational Exploration** through **Humans to Mars** activities

## Key Transit Habitat functions:

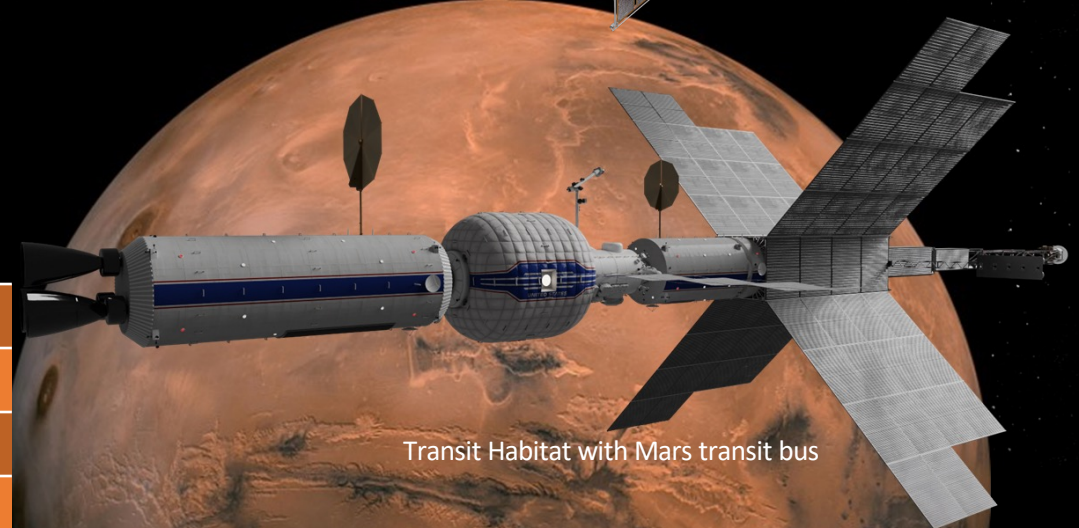
- Resource sharing and interfacing across multiple lunar and Mars architectures
- Radiation and Solar Proton Event (SPE) protection
- Docking capabilities
- Regenerative ECLS
- Logistics storage

## Transit Habitat Concept Design Characteristics<sup>1</sup>

Crew Size	4 crew
Mission Duration	Up to 1,200 days
Solar Orbital Range	0.6 – 1.6 AU
Lifetime	15 years
Mass Target	26.4 mt dry, ~50 mt Mars departure mass
Max Power	20 KWe
Pressurized Volume	~400 m <sup>3</sup>
Operating Pressure	10.2 – 14.7 psia (56.5-101.4 kPa)
Oxygen Composition	26.5-24.1% (limits for 10.2-14.7 psia)



Transit Habitat at Gateway



Transit Habitat with Mars transit bus

<sup>1</sup> Harris, Danny W., et al. *Moon to Mars (M2M) Habitation Considerations: A Snap Shot As of January 2022*. No. M-1538. 2022.



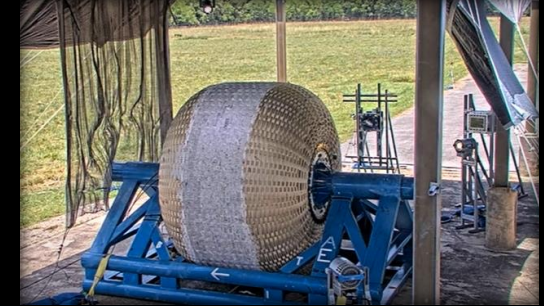
# NextSTEP Appendix A - Habitation



Advancing near-term habitation reference concepts through collaborative public-private partnerships

- Current contracted partners
  - Boeing
  - Lockheed Martin
  - Sierra Space
- Softgoods (inflatable) advancement
  - Burst testing:
    - Determine ultimate burst pressure of article
  - Creep testing\*:
    - Determine material deformation under load
    - Estimate operation life of structure
- Additional efforts:
  - Habitat concept design & maturation
  - Structural material trades & exploration

Sierra Space Burst Test Video - YouTube



NASA before/after images from test conducted at Johnson Space Center in collaboration with Sierra Space and ILC Dover



Lockheed Martin Burst Test Video - YouTube



\*2 Sierra Space creep tests completed outside of NextSTEP funding via reimbursable space act agreements.



# Habitation System Challenges

## Surface Habitat

- ❖ Limited spares/logistical resupply chain
  - Logistics storage capacity
- ❖ Limited maintenance & repair capability
  - Radiation protection
  - Dormancy periods (up to 3 years)
- ❖ Survive the night thermal/power
  - Power conservation
  - Energy storage solutions
- Delivery mass constraints
- Dust contamination

Common challenge

Common challenge

## Transit Habitat

- ❖ No spares/logistical resupply chain
  - Logistics (frozen & dry) storage capacity
- ❖ No maintenance & repair capability
  - Radiation protection
  - Dormancy periods (up to 3 years)
  - Impact on propulsion element size
- ❖ Communications delays/blackouts
  - Human health and performance for long duration microgravity missions
- ❖ Waste and trash management in transit/loiter

- ❖ Possible ISRU opportunity or in-situ service

Challenges faced by NASA's current habitation concepts may be opportunities for ISRU demonstration





# ISRU Opportunities for Habitation



## Logistics Reduction

- In-situ produced oxygen & water for ECLS supply/replenishment
- In-situ produced oxygen & hydrogen for fuel cell reactant replenishment



## Maintenance Support

- In-situ produced feedstocks for additive manufacture of spare/replacement parts
- Landing pads, roads, berms to mitigate plume effects & debris transport



## Radiation Protection

- Regolith-produced shielding materials for safe havens or fission power systems



## Survival Power

- Power generation and/or energy storage as a service
- Regolith-produced thermal barriers for sensitive materials



## Communications Support

- Disruption-tolerant relay systems as a service



Consumable <sup>1,2</sup>	Surface Habitat (4 crew, 28 days)	Transit Habitat (4-crew, 1110 days)
Oxygen (open-loop)	92+ kg	N/A
Oxygen (regenerative)	30 kg	123 kg
Water (open-loop)	1,139+ kg	N/A
Water (regenerative)	288 kg	182 kg

<sup>1</sup> Harris, Danny W., et al. *Moon to Mars (M2M) Habitation Considerations: A Snap Shot As of January 2022*. No. M-1538. 2022.

<sup>2</sup> Anderson, Molly S., Michael K. Ewert, and John F. Keener. *Life support baseline values and assumptions document*. No. NASA/TP-2015-218570/REV1. 2018.

Habitat outputs (packaging & human waste materials, unprocessed CO<sub>2</sub>, etc.) may also be available to support repurposing

ISRU pilot demonstrations may mitigate habitation challenges while building towards sustainable & profitable capabilities

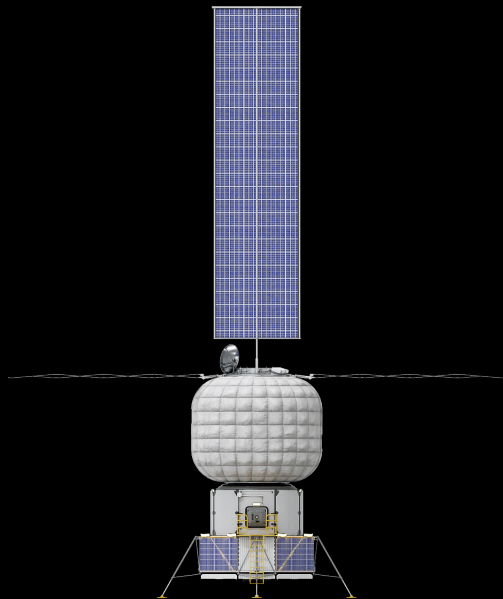
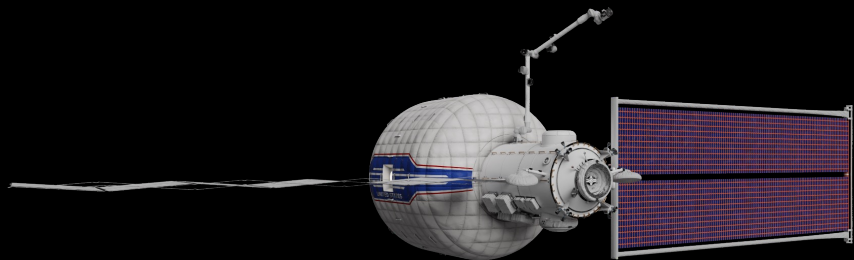
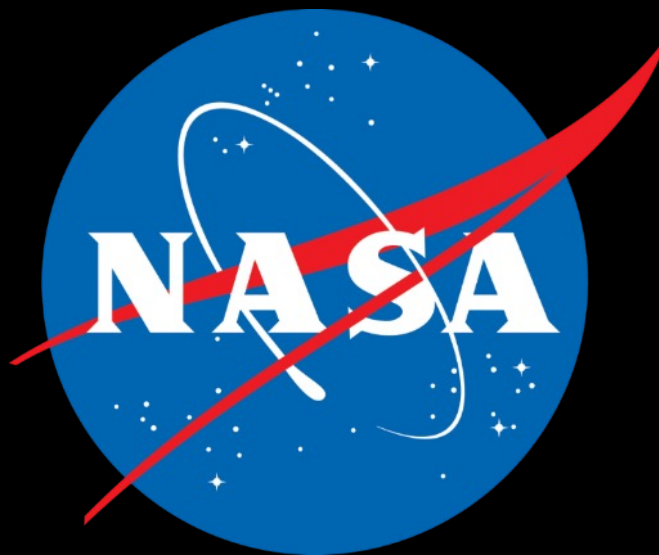


Image from ICON and SEArch+.



NASA.gov

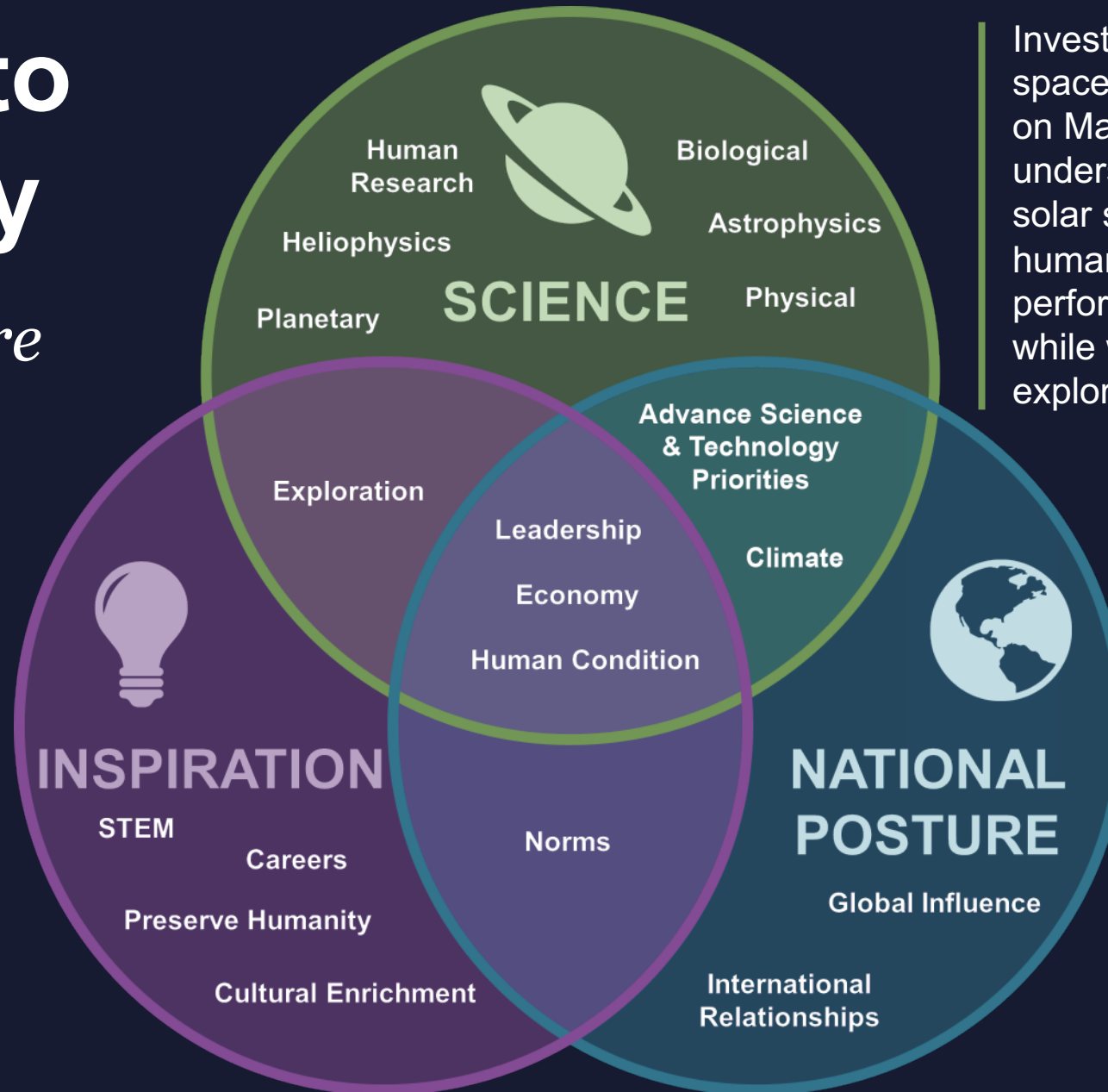


# BACKUP CHARTS

# Benefits to Humanity

## *Why We Explore*

Accepting audacious challenges and succeeding through perseverance and tenacity in the face of adversity motivates current and future generations to dare mighty things.

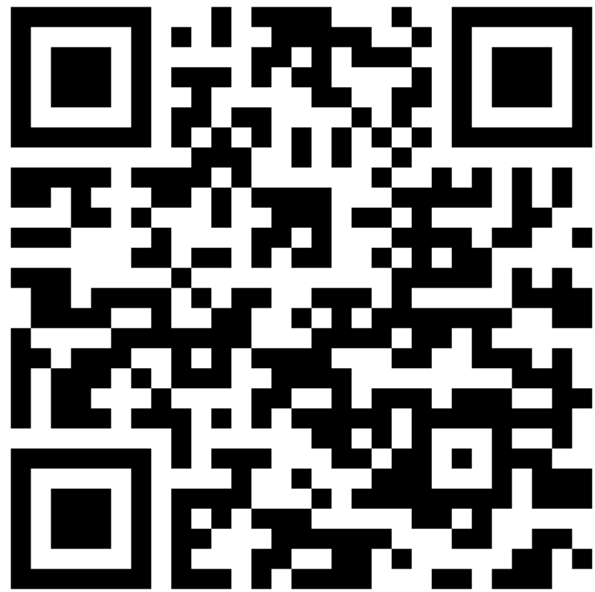


Investigations in deep space, on the Moon, and on Mars will enhance our understanding of the solar system, Earth, the human body, and how to perform new operations while we are out there exploring.

What we choose to do, how we do those things, and who we do them with greatly impacts our place in the world today, our quality of life, and our possibilities for the future.



# Architecture Concept Review Products



## Architecture Definition Document

Detailed documentation snapshot of NASA's human spaceflight architecture and exploration strategy



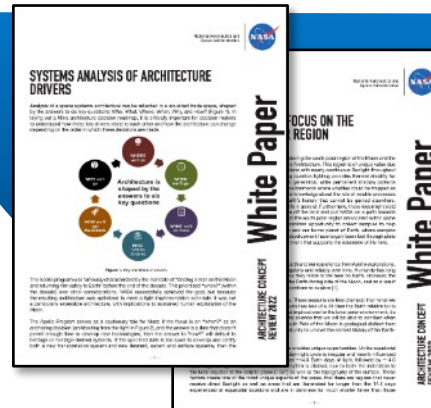
## Moon to Mars Architecture Summary

High-level overview of NASA's Moon to Mars architecture and exploration strategy



## White Papers

Six papers on architecture study details for frequently discussed topics



[www.nasa.gov/MoonToMarsArchitecture](https://www.nasa.gov/MoonToMarsArchitecture)



# Evolving Habitation Systems for SUSTAINABLE HUMAN EXPLORATION

Use ISS as Testbed for Evolution  
of ECLSS and CHPS



*International Space Station (ISS)*

Continue Testbeds on  
Commercial Platforms in LEO



*Notional Commercial  
Platform in LEO*

Infuse Technologies  
into Gateway

*Orion and  
Gateway*

- Toilet
- CO<sub>2</sub> removal
- Environmental monitoring
- Exercise technology
- Radiation protection and monitoring
- Medical system
- Fire suppression and cleanup

Infuse Full Long Duration Microgravity  
ECLSS and CHPS into Mars Transport



*Mars-class Transportation*

- Highly-reliable regenerative ECLSS from ISS demonstration
- Environmental monitors
- Exploration food system
- Countermeasures
- Medical system
- Radiation protection

Complementary Ground  
Tests and Analogs

- Food system performance and reliability testing
- CHPS integrated analogs



Human Landing System and Sustained  
Lunar Surface ECLSS-CHP Infusion

- Partial gravity and exploration atmosphere fire safety
- Exploration spacewalk pre-breathe and conops
- Surface habitat: regenerative ECLSS and CHPS adapted for surface
- Pressurized rover: ECLSS waste collection and transfer



Mars Surface ECLSS-CHPS

- Robust microbial and chemical monitoring
- Planetary protection compatible waste strategy

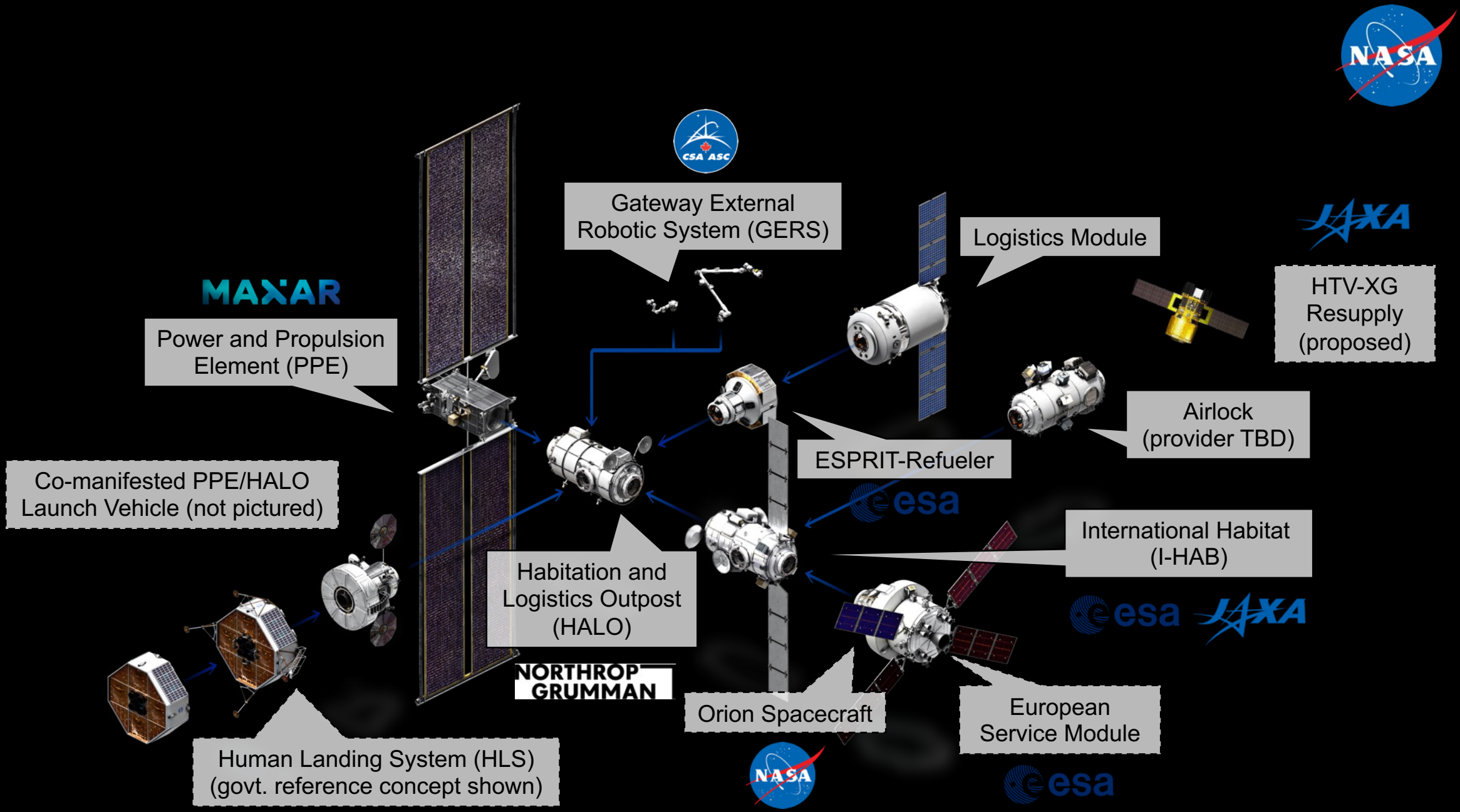


ECLSS = Environmental Control and Life Support Systems | CHPS = Crew Health and Performance Systems | LEO = Low-Earth Orbit

GRAPHICS NOT TO SCALE 20210116



# GATEWAY







## NextSTEP Appendix A: Habitation Systems



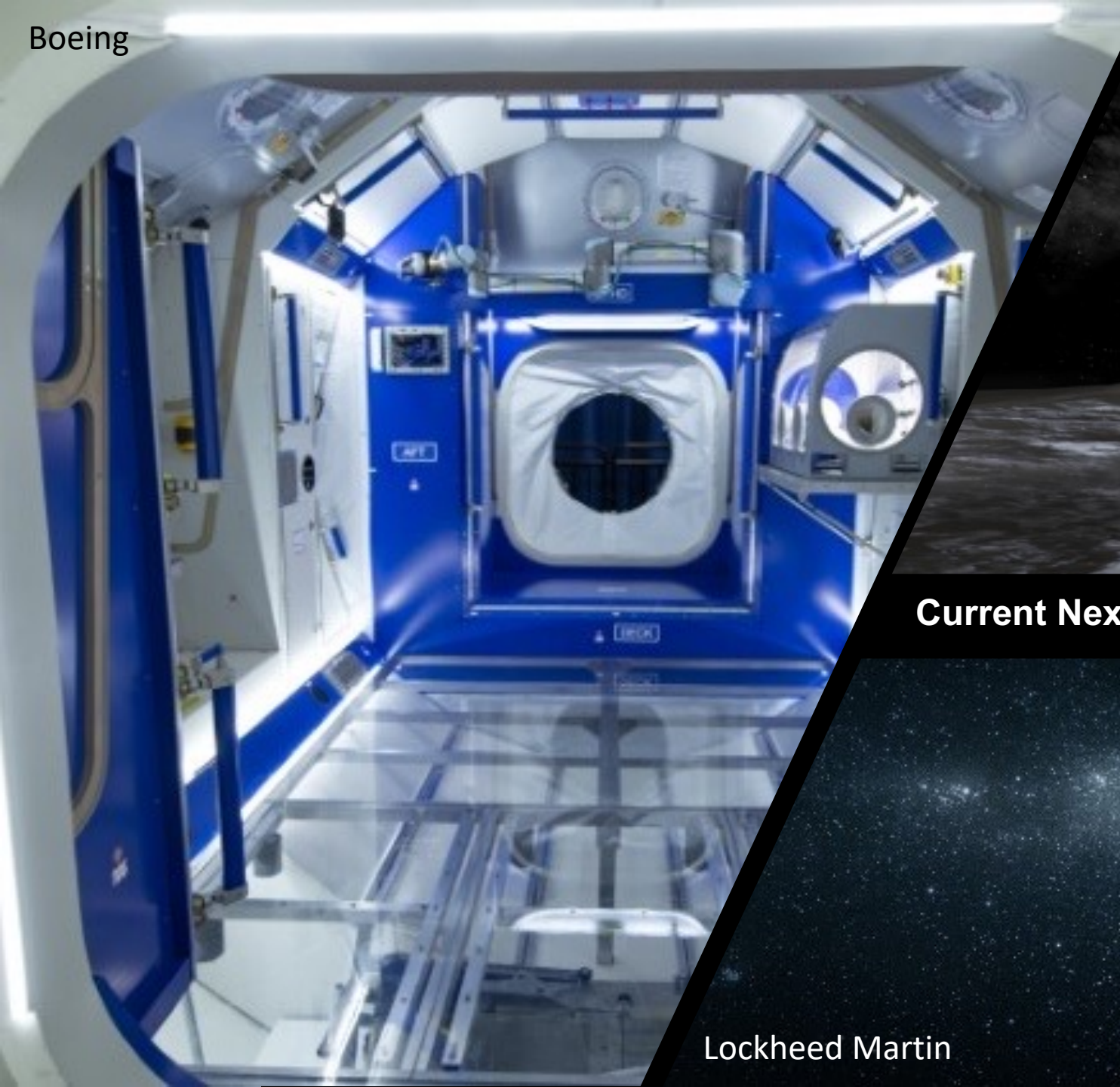
Phase 1 – 2015-2016: Concept Designs and Operations

Phase 2 – 2016-2019: Prototype Development and Testing

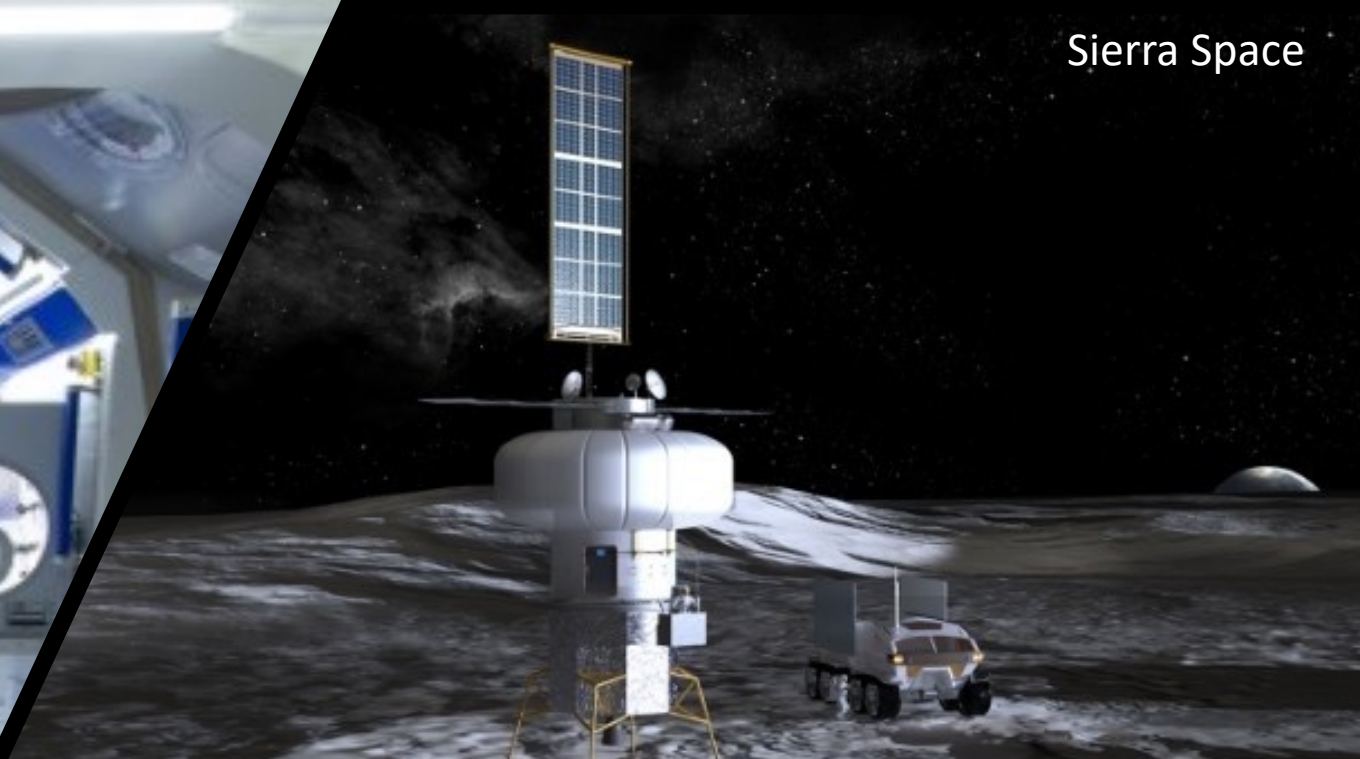
Phase 3 – 2019-2023: Softgoods maturation & long-duration, in-space, and surface habitat concepts



Boeing

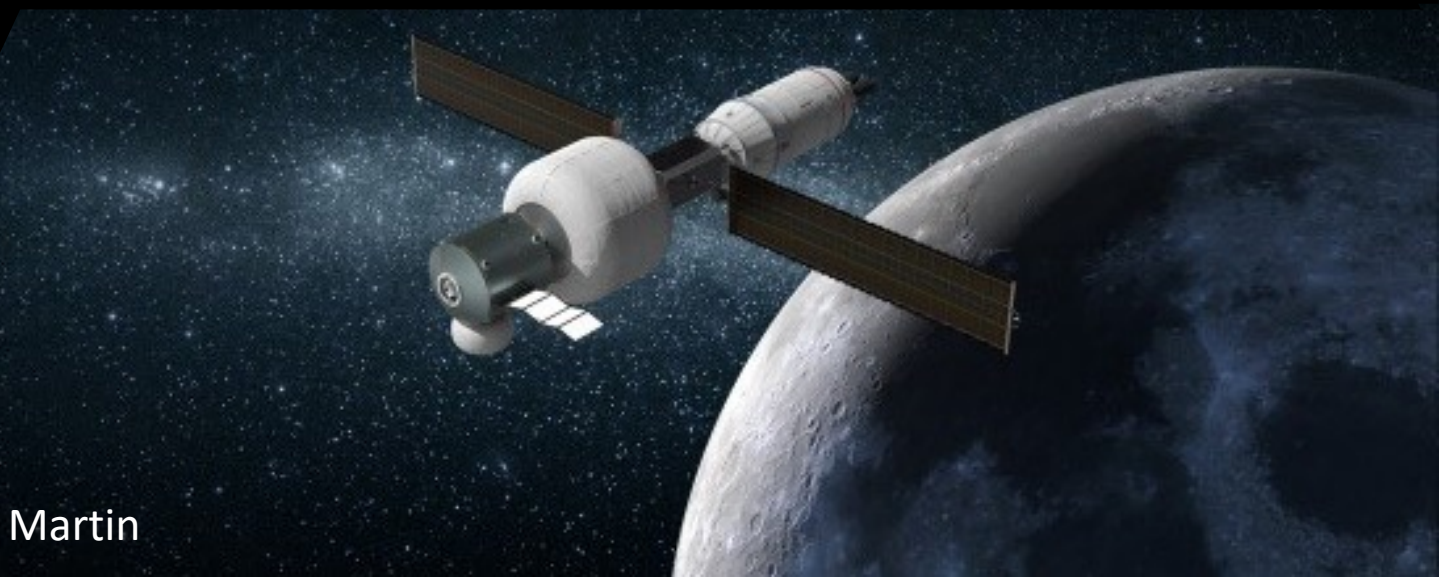


Sierra Space



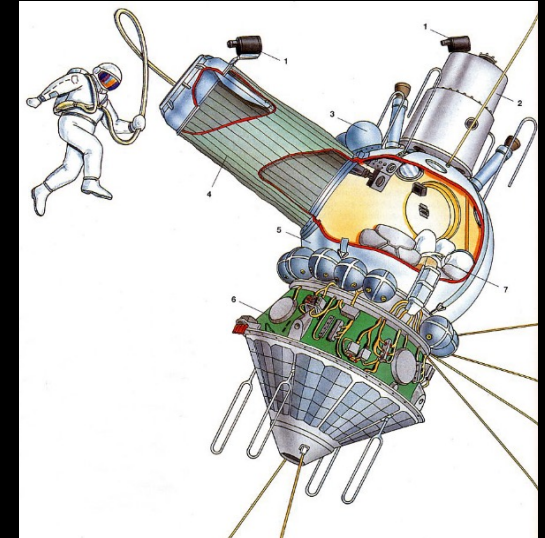
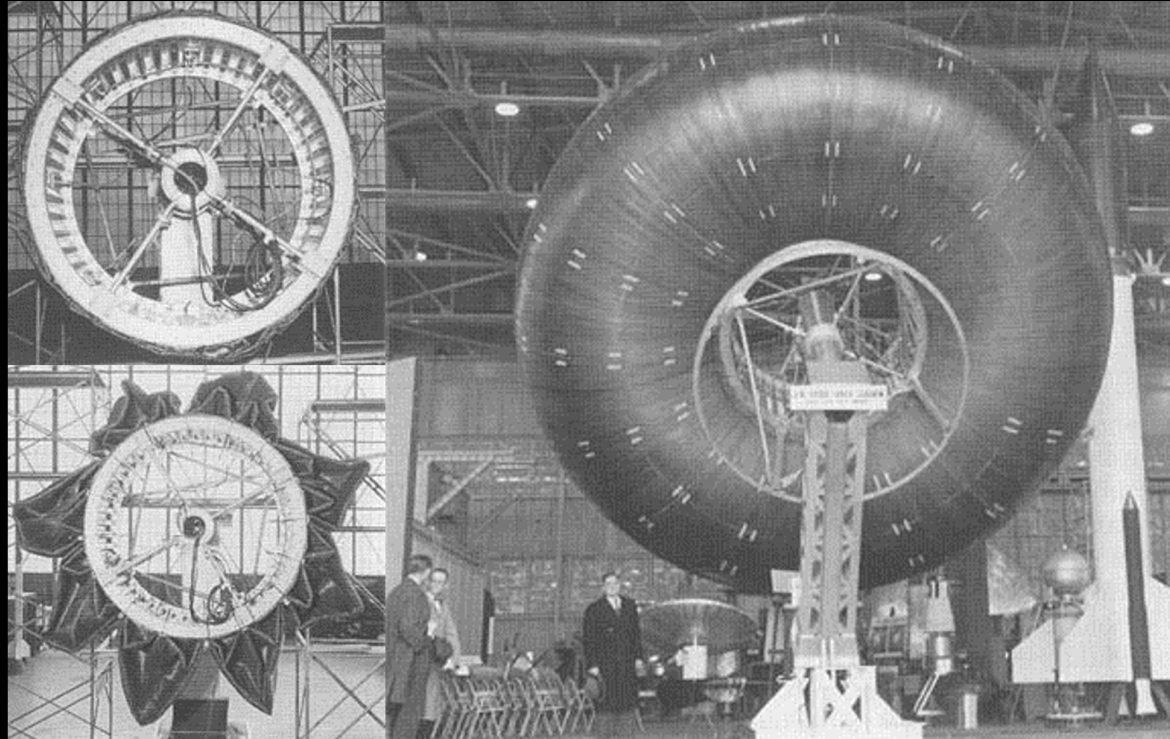
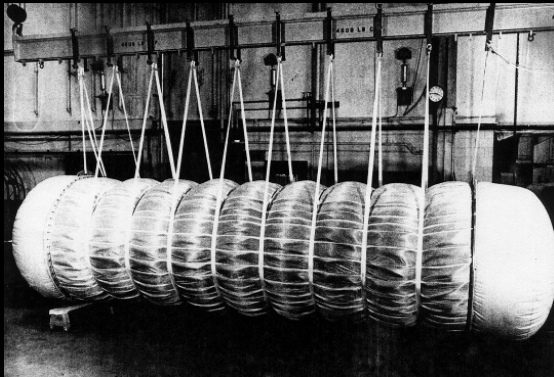
**Current NextSTEP Appendix A commercial partners**

Lockheed Martin



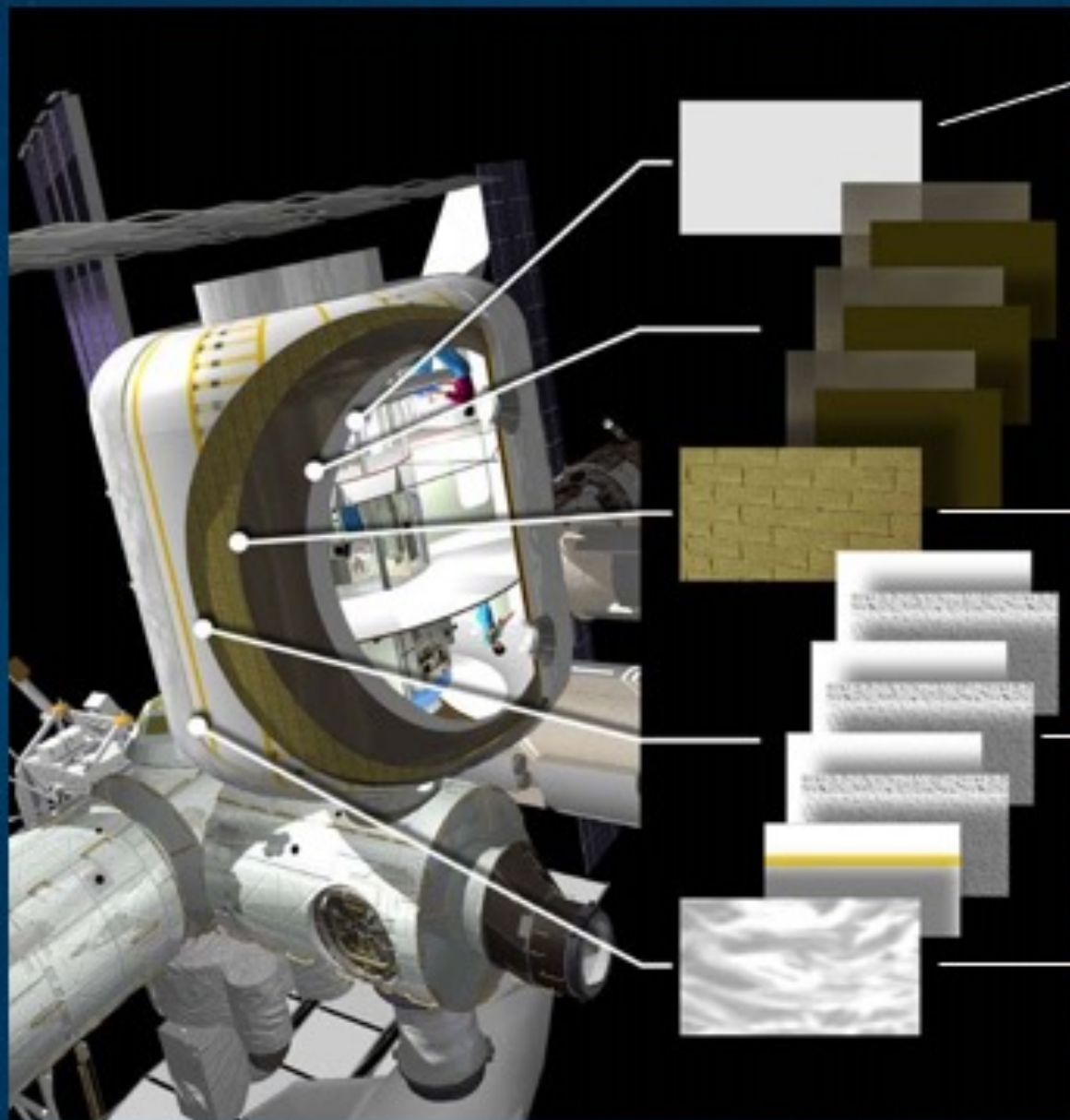


# History of Inflatables



USSR Voshod 2 Volga Inflatable Airlock (1964)  
(World's First EVA Was Out of An Inflatable Airlock)





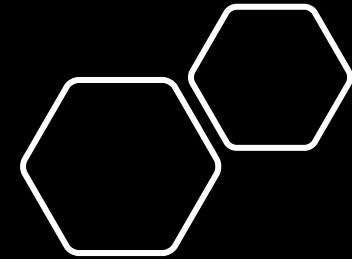
**Internal scuff layer:**  
Protects against  
punctures and fires.

**Bladder:** Keeps air  
inside the habitat  
from escaping

**Restraint Layer:**  
Main structure that  
helps hold the  
inflatable together

**Micro-meteoroid/  
Orbital Debris  
Shield:** Multiple  
layers protect the  
structure from  
impacts

**Thermal Blanket:**  
Protects the crew  
from extreme hot  
and cold  
temperatures

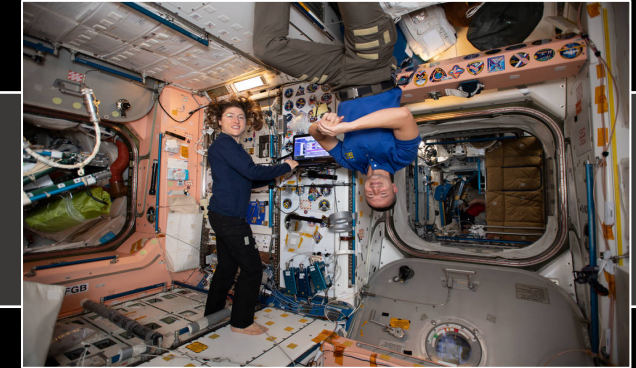


# *Inflatable Material System*

# Types of Space Habitats

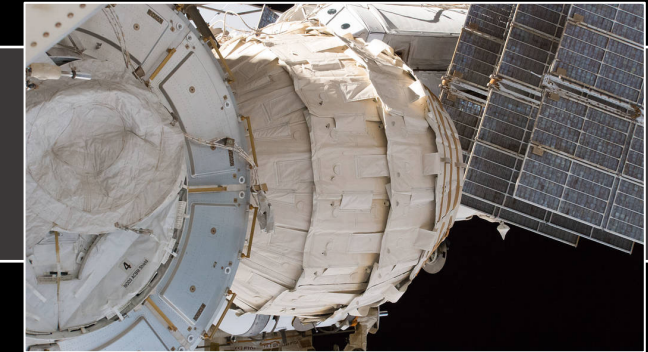
**A class I habitat** is a pre-integrated habitat, manufactured and integrated prior to launch

Example: ISS



**A class II habitat** is pre-fabricated prior to delivery, but deployed in space or on a planetary surface

Example:  
Inflatable ISS  
BEAM module



**A class III habitat** is built in situ using local resources

Example: 3D  
Printed Habitat on  
planetary surface





# Life Support Systems Provide Clean Air and Water

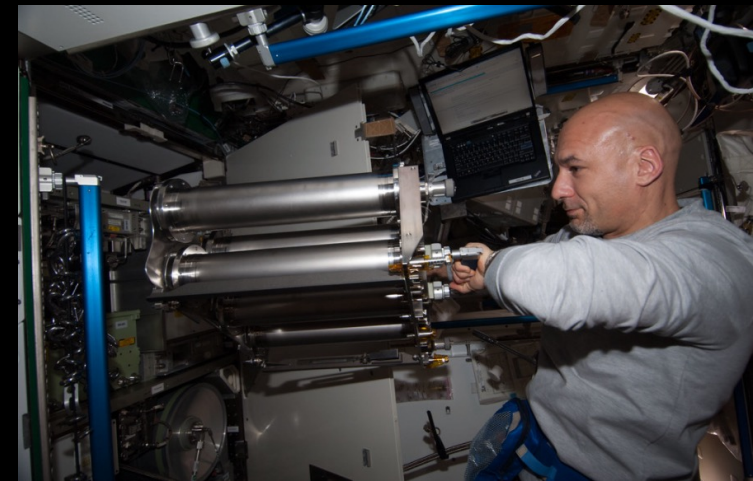


Two types of Environmental Control and Life Support Systems (ECLSS):

- Open loop ECLSS = provide all water, oxygen, and food from either stowed materials or cargo resupply
  - No recycling of outputs to inputs
- Closed loop systems process waste products into other resources and recover usable constituents
  - Closed loop ECLSS reduces dependence on resupply and imparts significant mass savings for future missions
  - Earth itself is a closed loop ECLSS system



Urine Processor Assembly



Water Processing Assembly



Oxygen Generation System



Carbon Dioxide Removal Assembly



# Human Needs on Space Missions

	Amount per crew member per day	4 crew on 30-day lunar surface mission
Food	2.4 kg (5.3 lbs)*	288 kg (635 lbs)
Drinking Water**	2.8 kg (6.2 lbs)	336 kg (744 lbs)
Water for food	0.5 kg (1.1 lbs)	60 kg (132 lbs)
Oxygen**	0.9 kg (2.0 lbs)	108 kg (240 lbs)
Clothing	0.2 kg (0.4 lbs)	24 kg (48 lbs)
Trash produced	1.7 kg (3.7 lbs)	204 kg (440 lbs)
CO2 produced**	1.1 kg (2.4 lbs)	132 kg (291 lbs)

\*about 15 percent of this weight is food packaging material \*\* assumes 82 kg astronaut for consumables planning (~180 lbs)

